



Near Earth Asteroid Scout Mission Update

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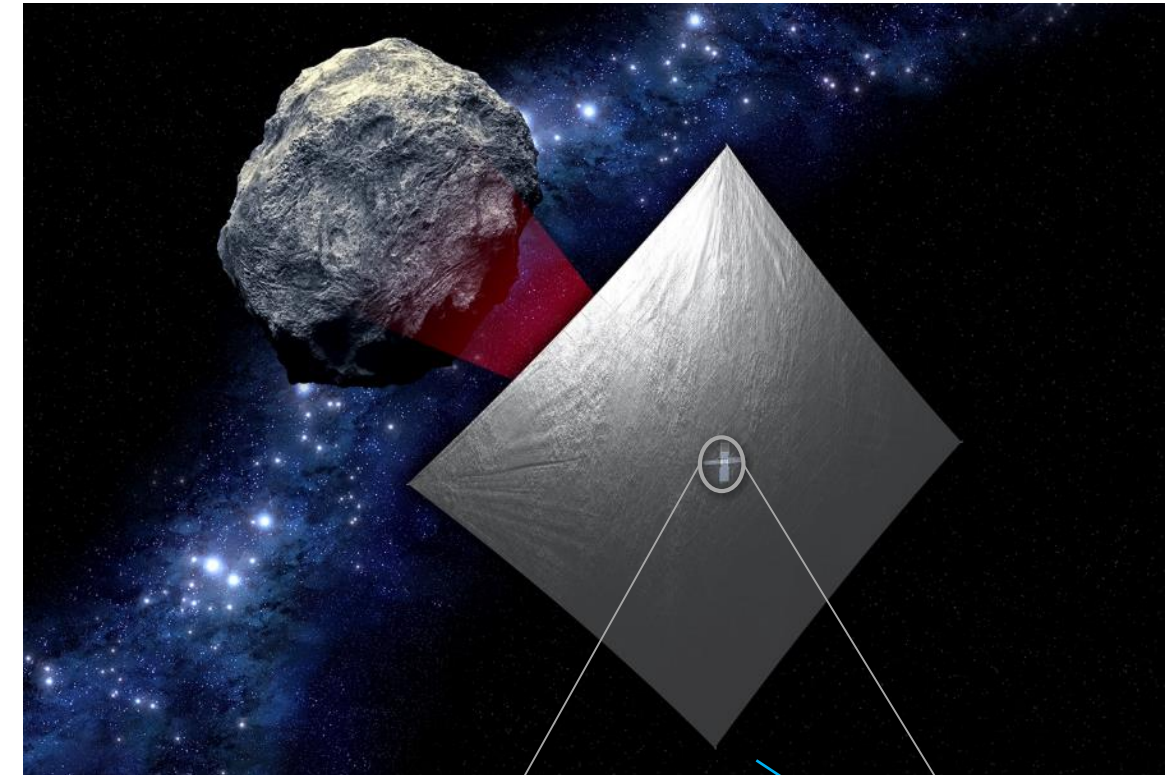
The Near Earth Asteroid Scout Will

- Image/characterize a NEA during a slow flyby (~m/s)
- Demonstrate deployment and navigation using a solar sail

Key Spacecraft & Mission Parameters

- 6U CubeSat
- ~86 m² solar sail propulsion system
- Manifested for launch on the Space Launch System Artemis 1, in 2021
- 1 AU maximum distance from Earth

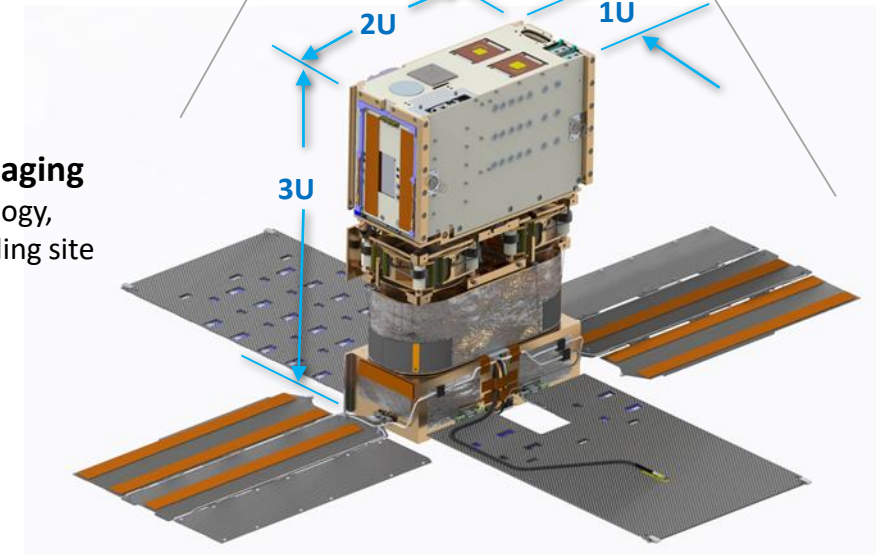
Leverages: combined experiences of MSFC and JPL with support from GSFC, JSC, & LaRC

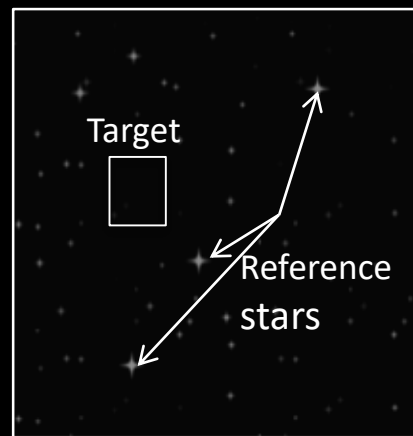


Target Reconnaissance with medium field imaging
Shape, spin, and local environment



Close Proximity Imaging
Local scale morphology, terrain properties, landing site survey

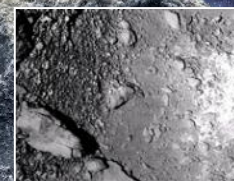
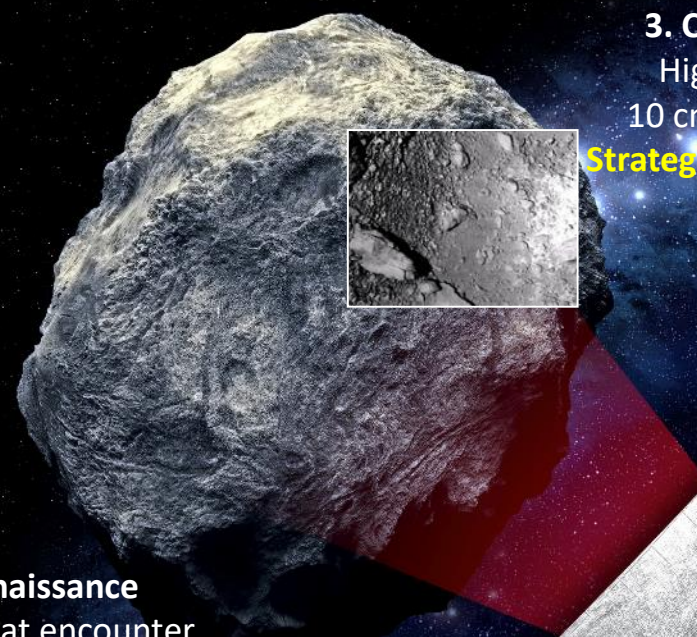




1. Target Detection and Approach:
50K km, Light source observation
SKGs: Ephemeris determination and composition assessment



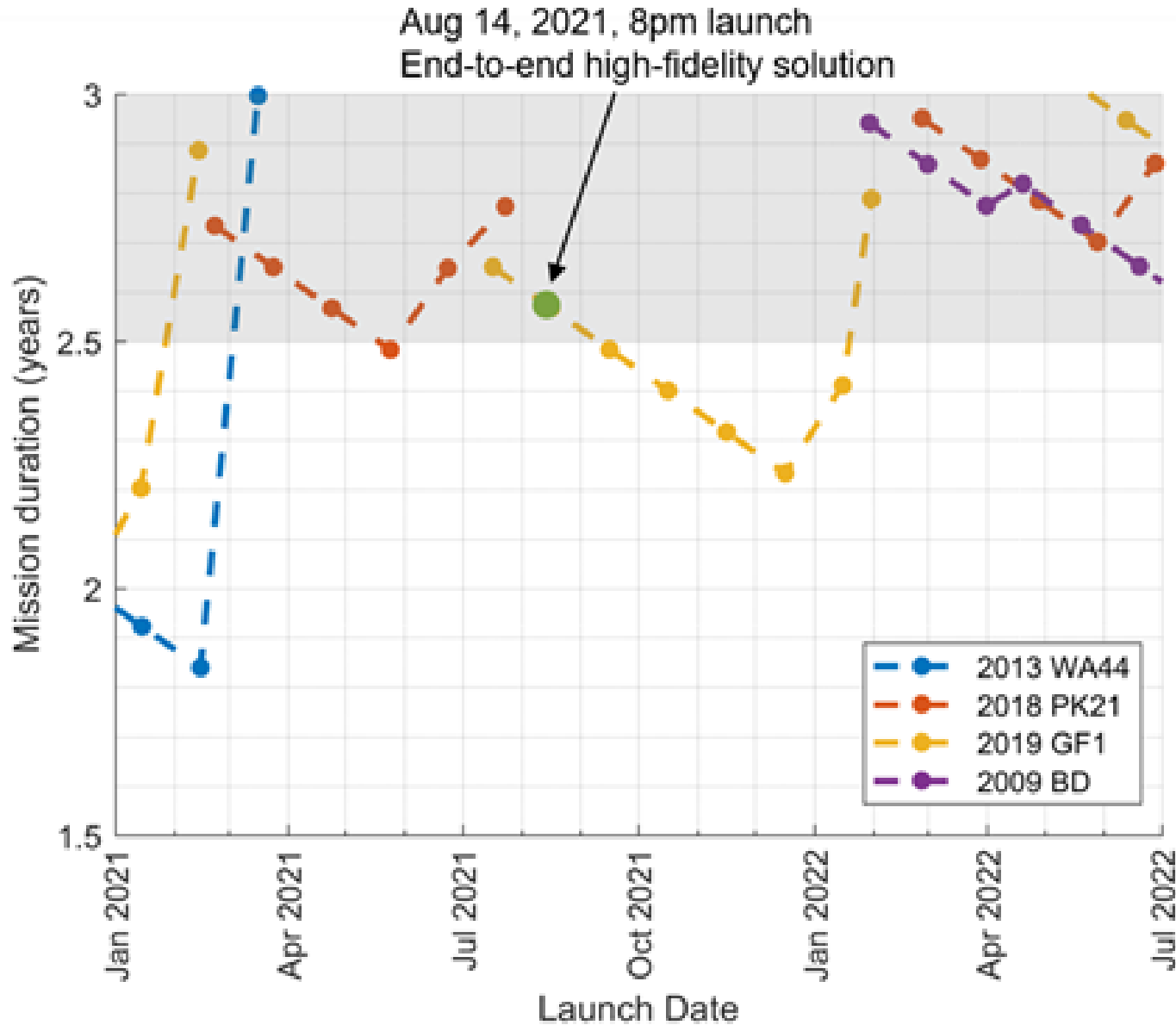
2. NEA Reconnaissance
<100 km distance at encounter
50 cm/px resolution over 80% surface
SKGs: volume, global shape, spin properties, local environment



3. Close Proximity Science
High-resolution imaging,
10 cm/px over >30% surface
Strategic Knowledge Gaps (SKGs):
Local morphology
Regolith properties

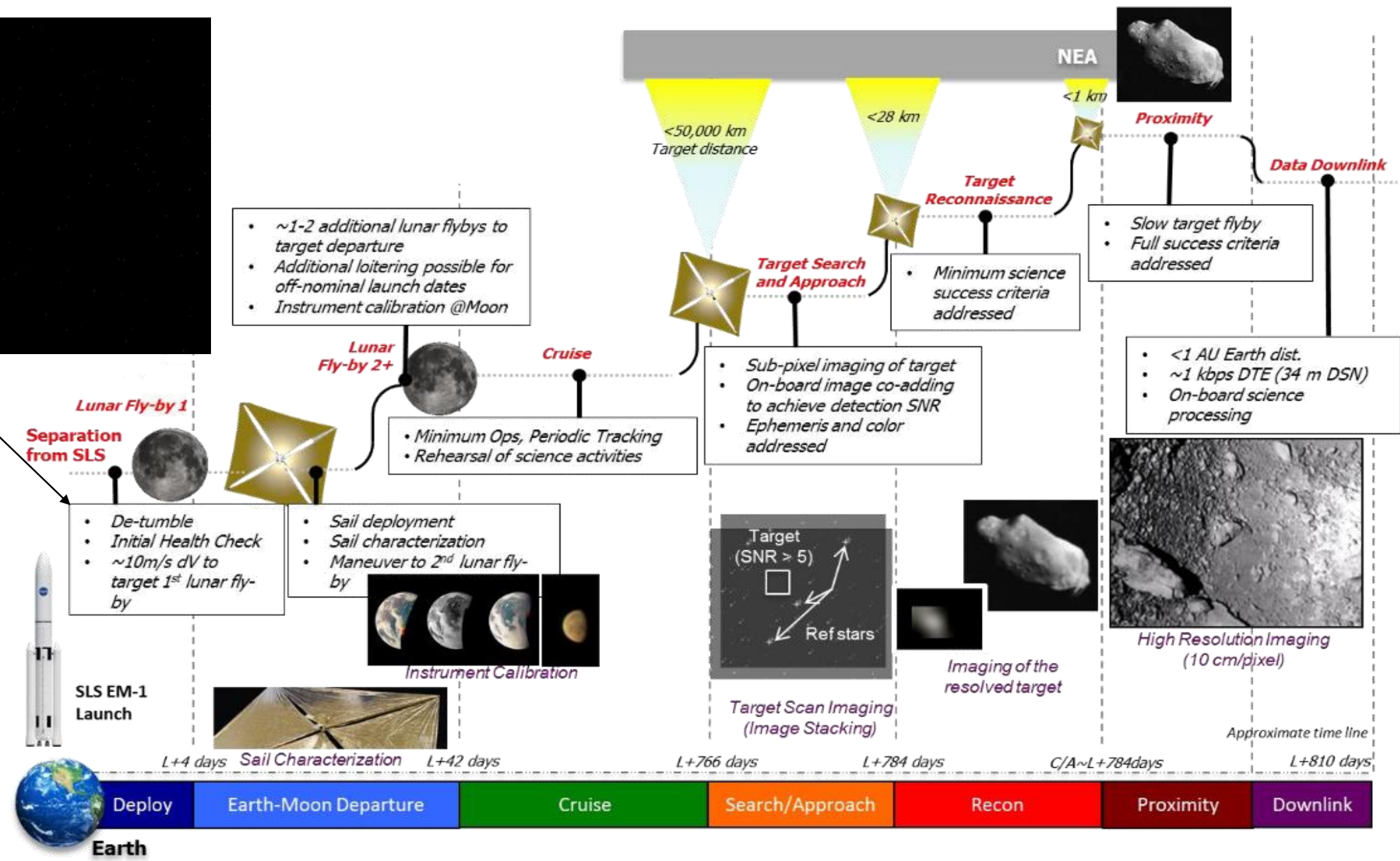
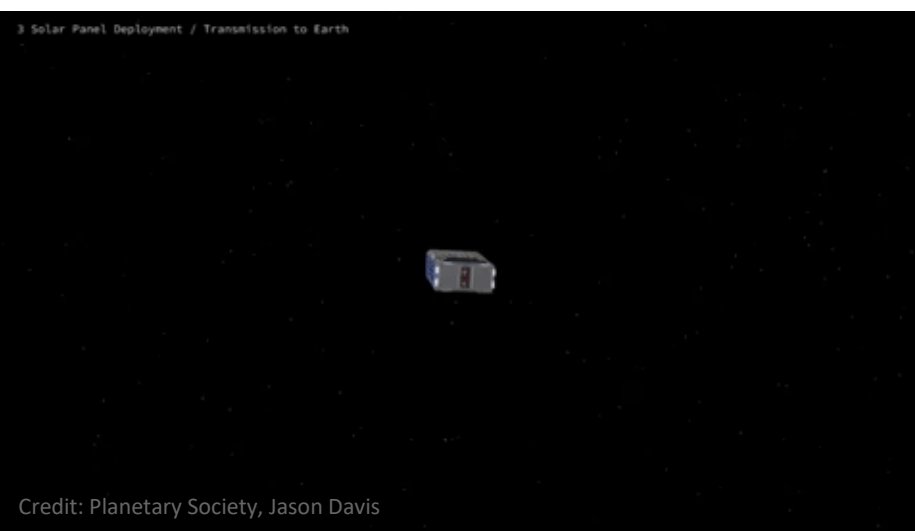


JPL

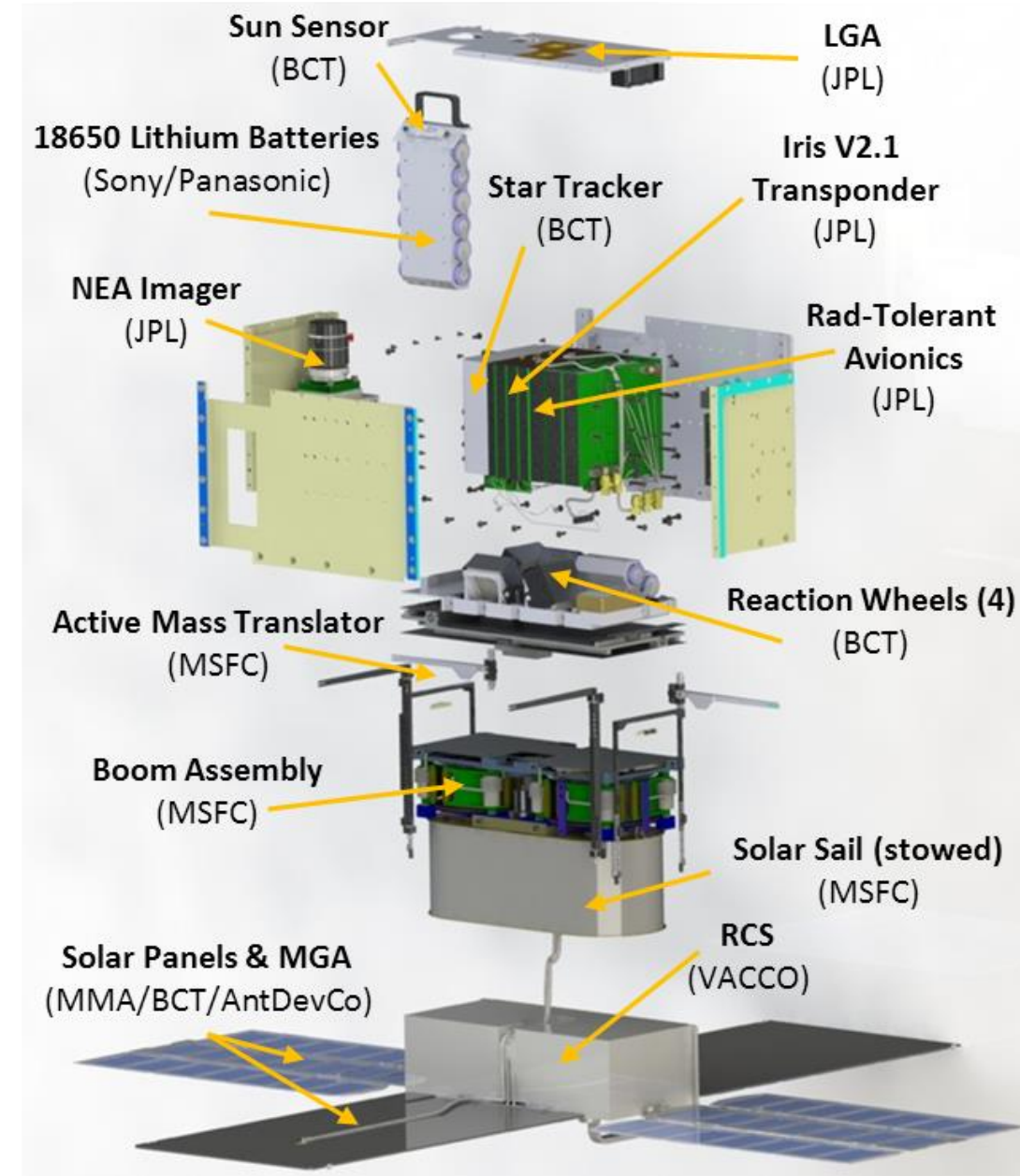


- Target changes with launch date and trajectory baseline from SLS
- Current target is 2019 GF1
- Available for launch dates from August 2021 thru February 2022

04/14/20
version



Payload	<ul style="list-style-type: none"> Updated OCO3 Context Camera
Mechanical & Structure	<ul style="list-style-type: none"> "6U" CubeSat form factor <14 kg total launch mass Modular flight system concept
Propulsion	<ul style="list-style-type: none"> ~86 m² aluminized CP-1 solar sail (based on NanoSail-D2)
Avionics	<ul style="list-style-type: none"> Radiation tolerant architecture
Electrical Power System	<ul style="list-style-type: none"> Trifold deployable solar arrays with GaAs cells (~51.2 W EOL at 1 AU solar distance) 6.2 Ah Battery 10 -12.3 V unregulated, 5 V/3.5 V regulated
Telecom	<ul style="list-style-type: none"> JPL Iris 2.1 X-Band Transponder; 4 W RF output power supports doppler, ranging, and D-DOR 2 pairs of INSPIRE-heritage LGAs (RX/TX) 8x8 element microstrip array MGA (TX); ~1 kbps to 34m DSN at 0.8 AU
Attitude Control System	<ul style="list-style-type: none"> 15 mNm-s (x4) Active mass translation system VACCO R-236fa (refrigerant gas) Reaction Control System Nano StarTracker, Coarse Sun Sensors & MEMS IMU for attitude determination





NEA Scout Approximate Scale



Deployed Solar Sail

Credit: NASA MSFC

Folded, spooled
and packaged in...

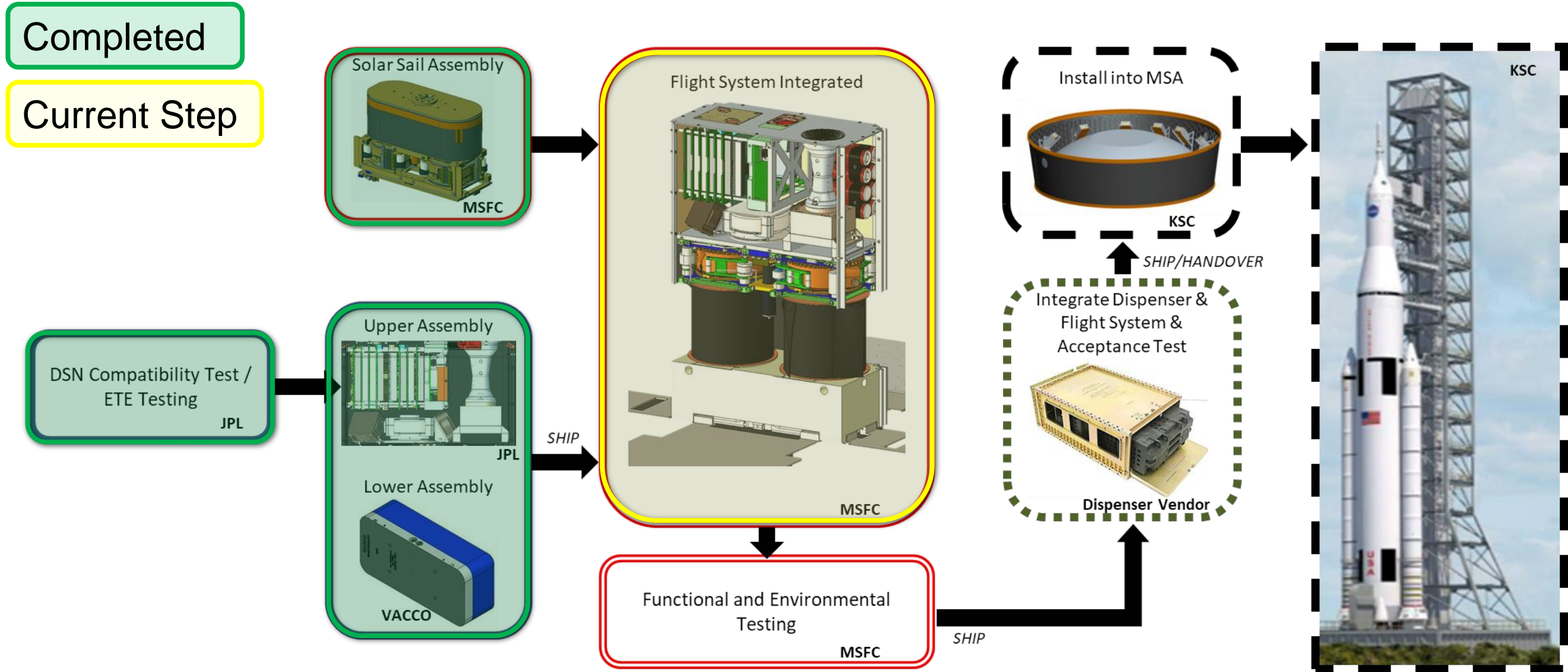
School Bus



Human

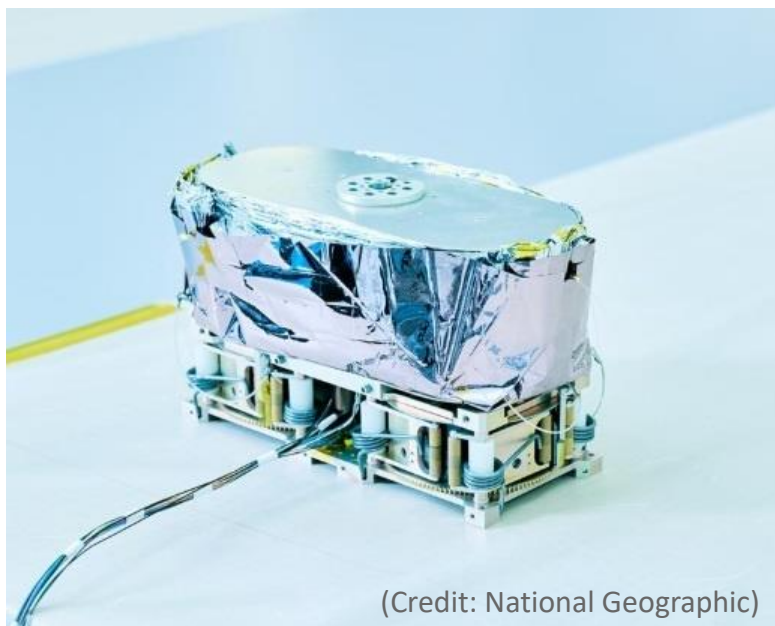
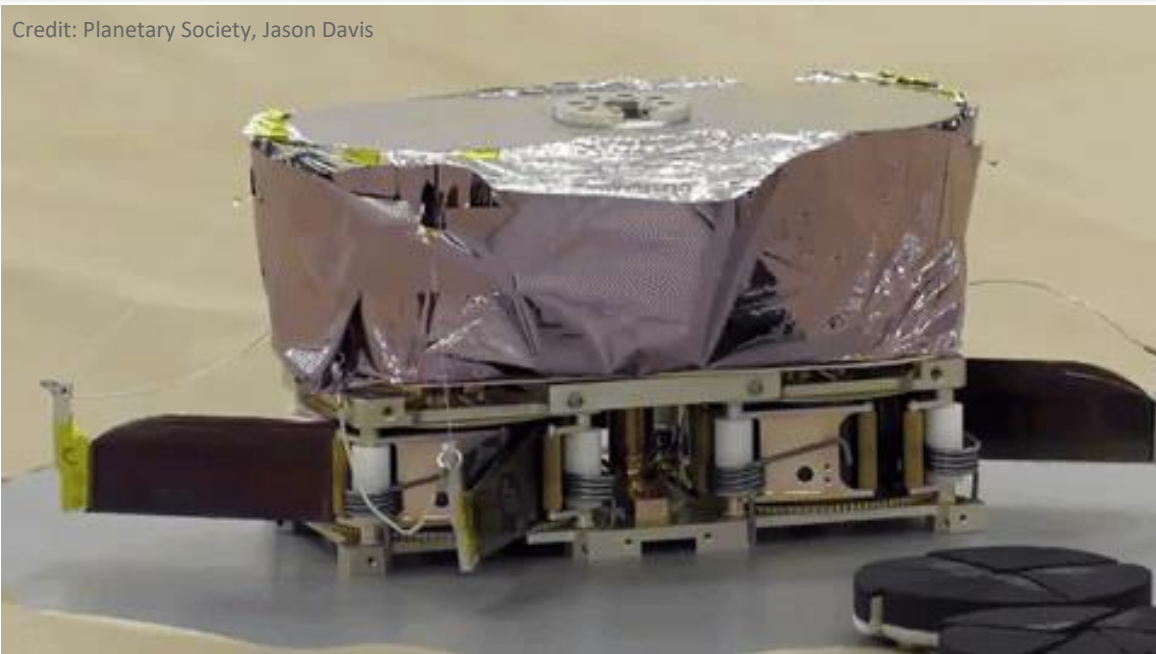


6U Stowed Flight System



Solar Sail Construction and Deployment

Credit: Planetary Society, Jason Davis

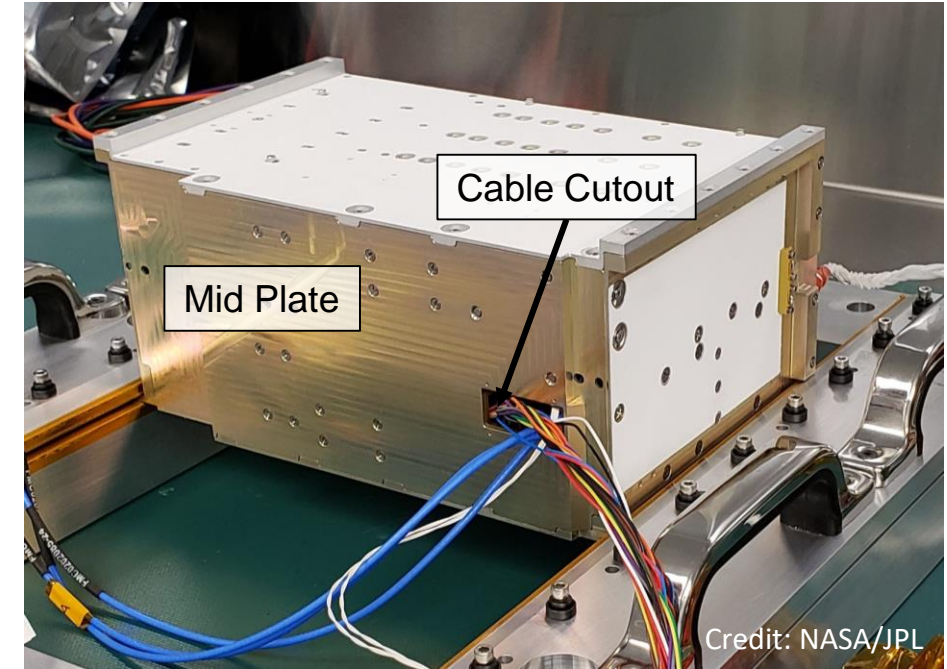
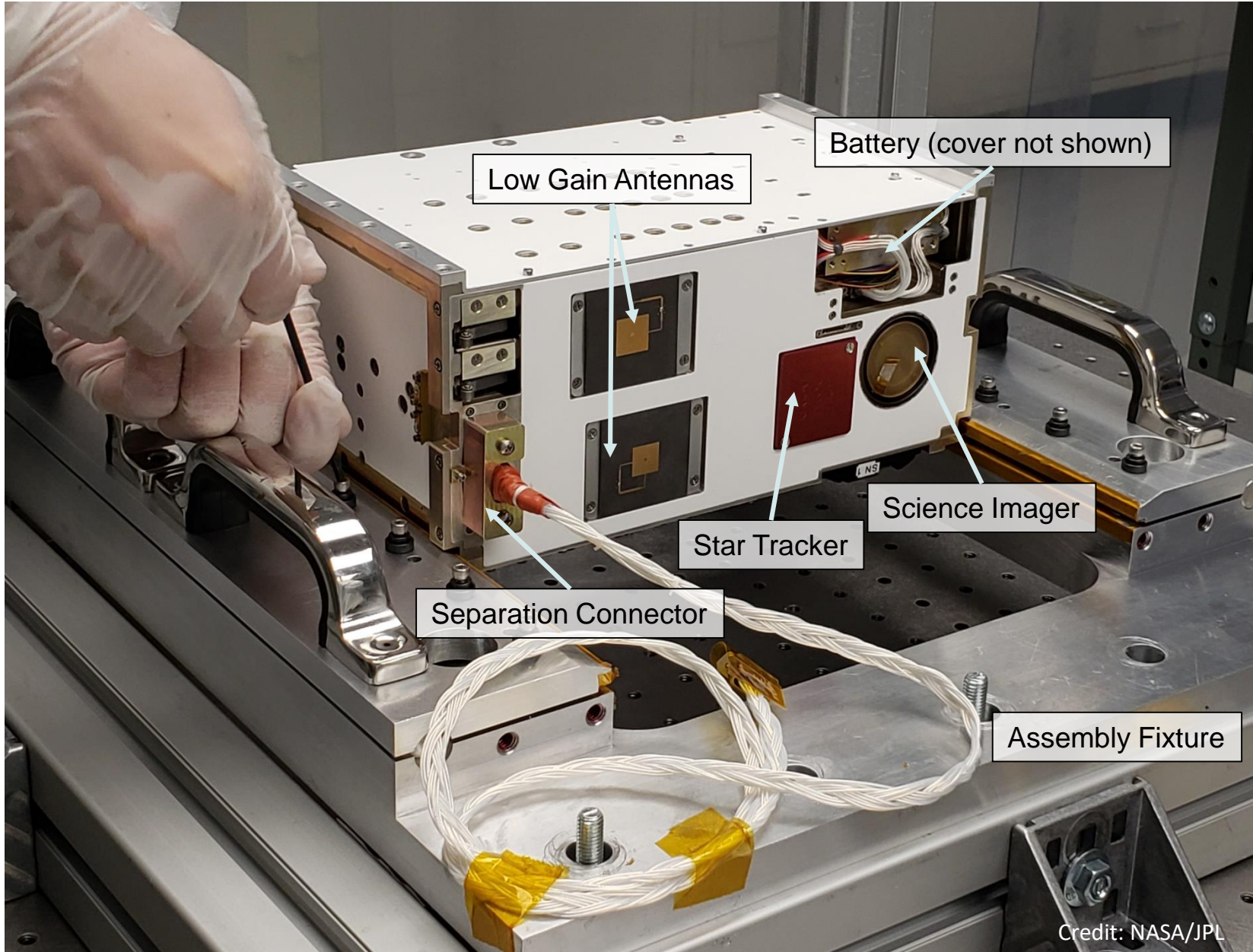


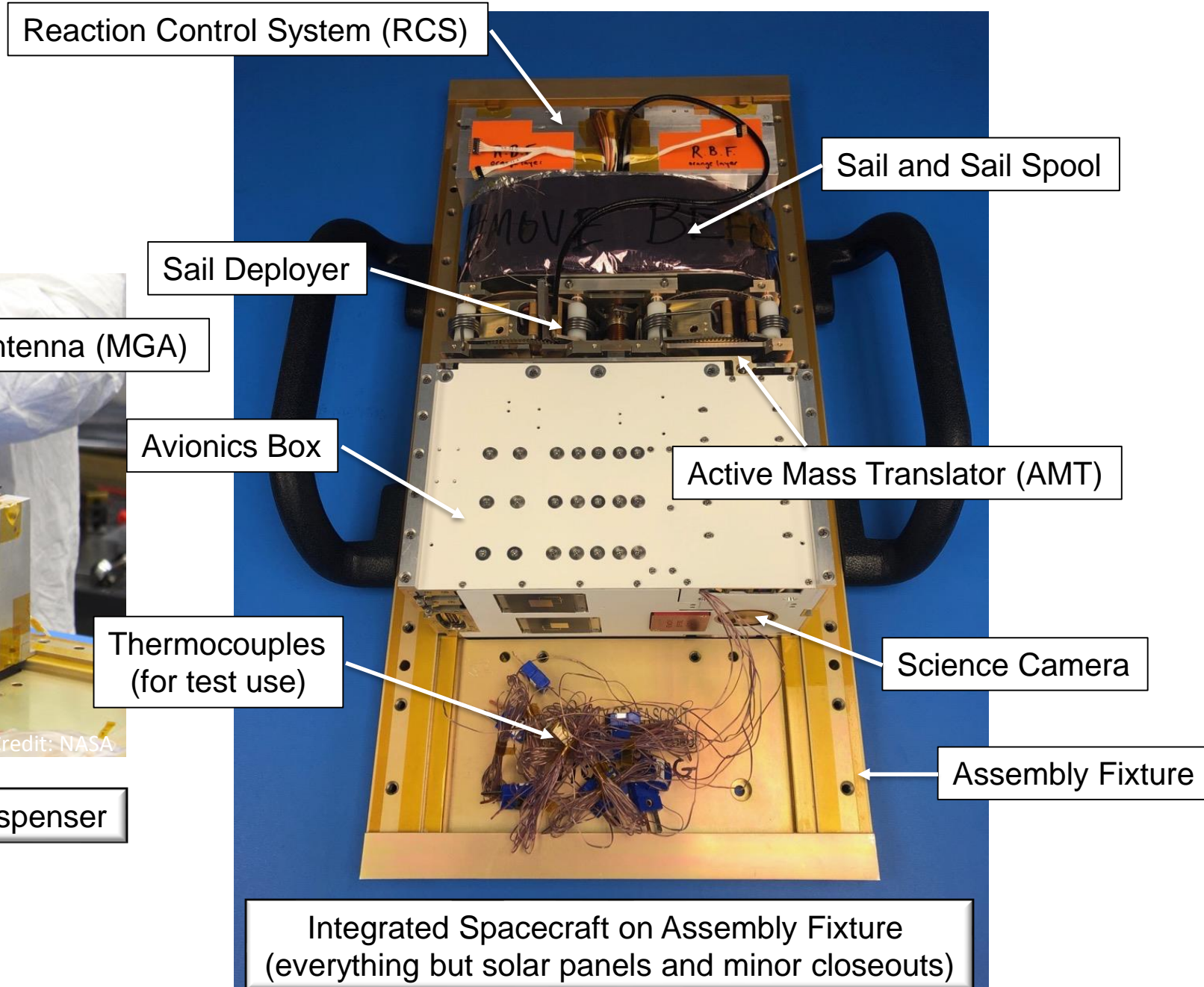
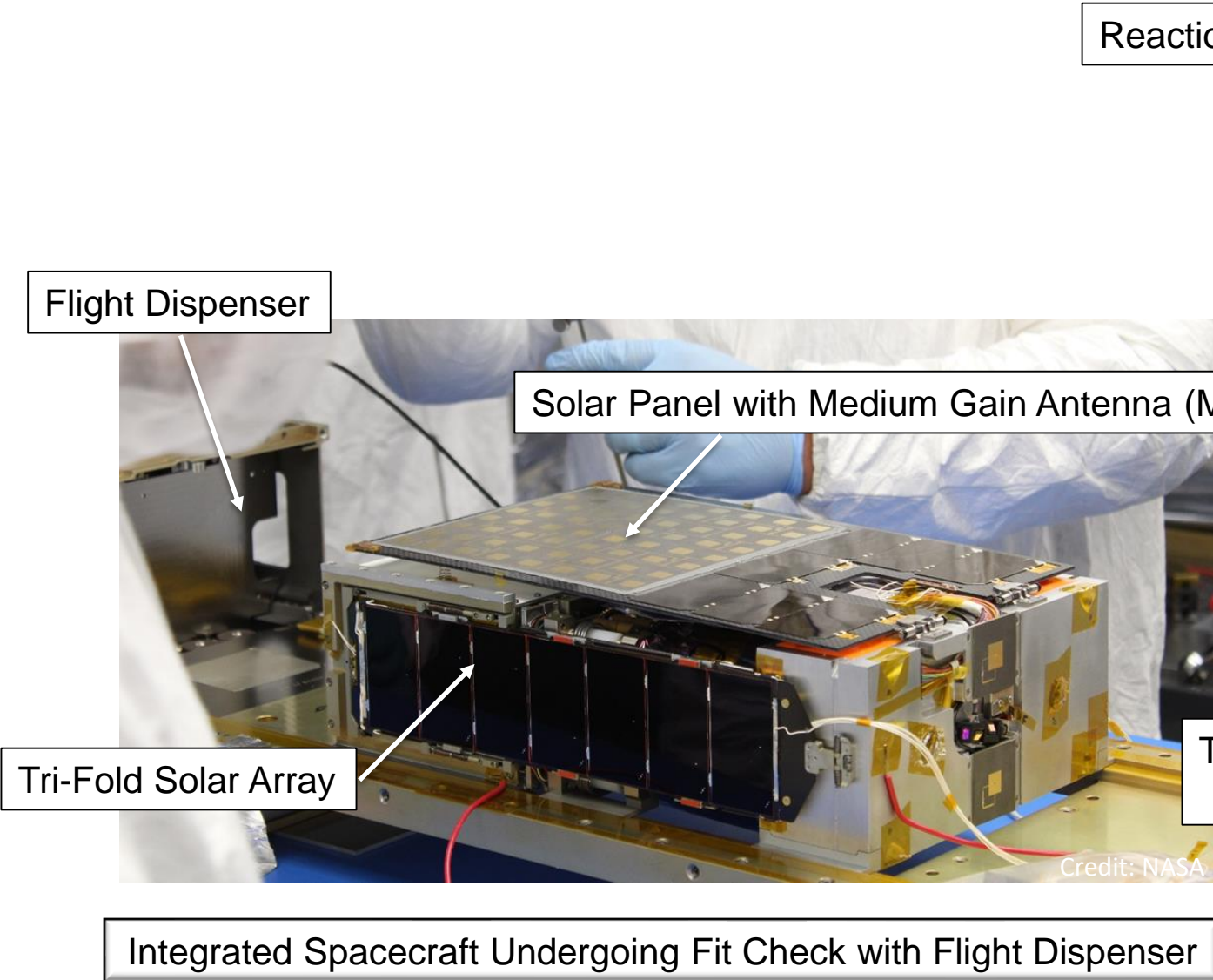
(Credit: National Geographic)



(Credit: National Geographic)

Avionics Box Delivery – August 2019







NEAS Status Summary



- Spacecraft integration ~95% complete, delayed by pandemic
 - Awaiting final installation of solar panels
- System level testing planned to begin once integration is complete
 - Functional
 - EMI/EMC
 - Vibration
 - Thermal
 - Day in the Life
- Delivery to SLS/ Tyvak for flight No Earlier Than end of Sept 2020



Questions?

